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# Measurement of the structural damping ratio of some common materials at 293 K and 77 K in air

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Electron-Ion Collider **Brookhaven National Laboratory** 

# **U.S. Department of Energy**

USDOE Office of Science (SC), Nuclear Physics (NP)

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# Measurement of the structural damping ratio of some common materials at 293 K and 77 K in air

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**Abstract.** The structural damping ratio of various materials has been measured at 293 K and 77 K through a rod bending excitation setup. The aim was to extract the material-dependent structural damping ratio as a first approach to in-vacuum damping. However, for simplicity, testing has been conducted in air with measures to minimize the effect of viscous damping from viscous drag. The setup and methodology are described in this paper, as well as the setup limitations and proposed ways to overcome these limitations for an improved setup. The results are presented in a summary table and discussed.

### 1. Introduction

A general trend in new particle accelerators projects is the reduction of transverse beam size, as this is useful to improve brightness for synchrotron light sources or luminosity for colliders. However, one downside is that it makes beam orbit stability – and so hardware position stability - proportionally more important [1][2]. To design particle accelerator supports with high stability, engineers typically work on limiting low-frequency resonances. But when that is not possible, or when a high stability is required up to a high frequency, some structural resonances can still be excited, and the resulting vibration amplitude needs to be assessed and put under control. The evaluation of the displacement triggered by these resonances requires the knowledge of the damping characteristics of the material being deformed. While in air the effective damping is a combination of material-dependent structural damping and viscous damping from air drag, in vacuum, only the structural damping component is present. In an attempt to approach values of structural damping values in this study, we tried to minimize the effect of air viscous damping. While absolute damping values reported still include some amount of viscous drag damping, relative values are useful to show variations between materials in an identical test environment. This study offers a determination of materials structural damping characteristics at room temperature (293 K) and around liquid nitrogen temperature (77 K) in air with efforts made to minimize air viscous damping.

### 2. Experimental methodology

# 2.1 Test setup

The test setup relies on measuring the vibration amplitude decay of a bending beam in a fixed-free configuration. Samples are  $\emptyset 0.25$  inch ( $\emptyset 6.35$  mm) rods with length of 8.5 inch (215.9 mm) for all samples tested both at 293 K and 77 K and length 12 inch (304 mm) for all samples tested excusively at 293 K.

The samples fixed-ends are clamped in a double-screw clamp in an attempt to limit the friction within the clamp assembly which would create artificial damping. When testing at 77K, a tubular

PTFE sleeve is fixed around the sample – without contact - to stratify the air and helps limit thermal convection and icing on the sample. Icing on the sample exposed surface has a potential to increase damping artificially by creating friction between the ice crystals formed.

For stiff samples, weights are clamped at the free end to reduce the bending oscillation frequency and so limit the air viscous damping. All samples were set to have a resonnance frequency between 10 and 30 Hz.

The samples are excited by loading up elastic energy, and releasing this energy suddenly to measure vibration decay. An accelerometer placed on the sample free-end records the acceleration amplitude. The accelerotmeter acquisition is set to 500 Hz, an order of magnitude above the resonnace frequency. Measurements are made from an acceleration of 2g (19.62 m/s²) down and damping ratio are calculated across as many periods as possible.

When tested at 77K, sample are immersed in LN2 until thermal equilibrium is established and then promptly put upright and excited. Some temperature uncertainty stems from the time needed to mount the sample and excite it – although this was done fast and results were usually repeatable - and some additional uncertainty come from the higher damping of cold and dense air and limited amount of freezing on the sample surface.

### 2.2 Data processing

An example of acceleration decay from the titanium sample is depicted on Figure 1.

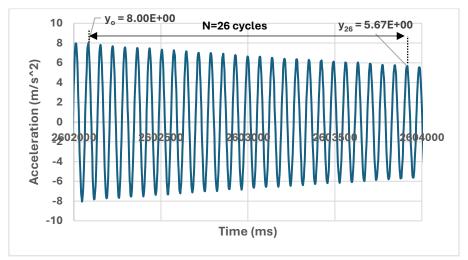


Figure 1 Acceleration decay from the vibration of the titanium alloy TiAl4V rod – resonance @14.36 Hz

A fast fourier transform is used to determine the precise resonance frequency, here 14.36 Hz. As described in Ref.[3] (Eq. 23), the damping ratio can be determined:

$$\xi = \frac{C}{C_0} = \frac{\ln(y_0/y_N)}{\sqrt{(2\pi N)^2 + \ln(y_0/y_N)^2}}$$

In this case, using values from Fig. 1:

$$\xi = \frac{\ln(8.00/5.67)}{\sqrt{(2\pi.26)^2 + \ln(8.00/5.67)^2}} = 0.0021 (0.21\%)$$

# 3. On the significance of the damping ratio $\xi$

The analysis of a 1D harmonic oscilator driven by a sinusoidal displacement excitation  $x_a(t) = Asin(\omega t)$  gives Ref. [3,4,5]:

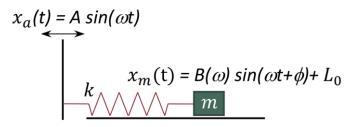


Figure 2:1D harmonic oscillator driven by a displacement – Figure from [5]

$$|B(\omega)| = \frac{|A|}{\sqrt{\left(1 - \left(\frac{\omega}{\omega_0}\right)^2\right)^2 + 4\xi^2 \left(\frac{\omega}{\omega_0}\right)^2}}$$

Note that the damping ratio also shifts the effective resonance frequency  $\omega_{res}$  from the undamped resonance frequency  $\omega_0$  as such:

$$\omega_{res} = \omega_0 \sqrt{1 - 2\xi^2}$$

So, at the damped resonance frequency  $\omega_{res}$ :

$$|B(\omega_{res})| = \frac{|A|}{2\xi\sqrt{1-\xi^2}}$$

In practice,  $\xi^2 \ll 1$  ( $\to \sqrt{1-\xi^2} \approx 1$ ) and so the vibration amplification at resonance  $\omega_{res}$ :

$$\frac{|B(\omega)|}{|A|} \propto \frac{1}{2\xi}$$

From this, to accurately predict the amplification when a structure resonance is excited, the knowledge of the damping ratio  $\xi$  is required, as the resonant displacement amplitude in inversely proportional to  $\xi$ .

In some applications, the integrated motion amplification over a frequency range is more useful than the peak resonant amplitude at given frequency. For example, this is true for the evaluation of accelerators components stability under a broadband excitation.

A discussion of getting the integrated amplification factor is offered in annex 1, the dependence of the integrated motion amplification is shown to be:

$$\int \frac{|B(\omega)|}{|A|} df \propto \frac{1}{\xi^{0.23}}$$
 (approximately)

For a consistent broadband excitation.

### 3. Test results and discussions

### 3.1 Results

Figure 3 summarizes the results of the testing described in section 2.

The damping ratio is represented as blue bars for 293K testing and orange bars for  $\sim$ 77 K testing. The error range represent the difference between the average value reported and the extreme value obtained in the multiple testing attempts for each sample.

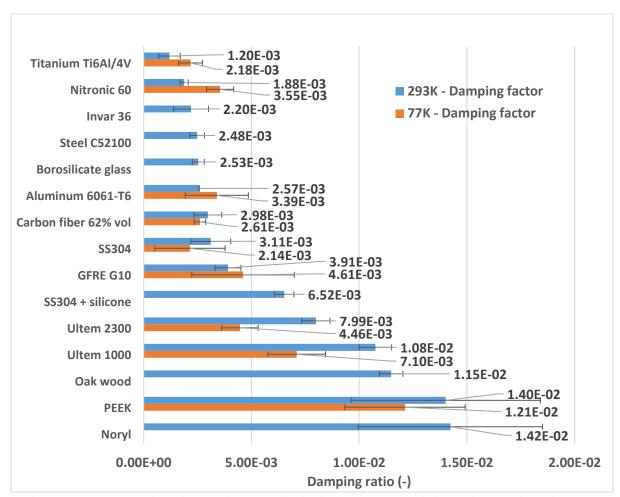


Figure 3 Structural damping ratio for various materials in air – samples ranked in increasing damping ratio at 293K

### 3.2 Discussion of the result

The data obtained are generally consistent with data available in the literature [6][7].

From Fig.3 metals and glass show the lowest damping ratio, followed by composite materials, wood and plastics. Available material certificates are available in appendix 2.

Short glass fibers in ULTEM tend to decrease the damping ratio (ULTEM 2300 has 30% short glass fibers compared to the unfilled 1000).

The damping ratio of plastics decreases markedly at 77K (Ultem 1000, 2300, PEEK). This is consistent with results previously reported in [8]. However, this effect is much less present in the long fiber-composite samples tested - G10 and carbon fiber CFRE - where the damping ratio change is relatively small between 293 K and 77 K.

Metals also show relatively small variation. While Ti-6Al-4V, Nitronic and Al 6061-T6 seems to show a minor increase in damping at cold, stainless-steel 304 shows a slight decrease.

However, with the relatively low values of damping ratio of metals, it is hard to rule out an effect of higher viscous damping due to the dense cold air around the sample or local freezing on the sample surface at cold temperature. To refine these results further, one should aim to use a setup under vacuum as discussed later in this report.

### 3.3 Applying an elastomer coating to improve damping?

Elastomers are well-known to have a superior damping coefficient [7].

In order to evaluate the effect of applying an elastomer coating on the surface of a metallic sample, the stainless-steel SS304 sample was coated with a thin layer of "self-levelling" silicone (Dow Corning 734®). The sample was then wiped immediately after silicone application to obtain a very thin and consistent coating thickness.

Testing at room-temperature showed an interesting increase in damping ratio of the stainless-steel sample from 3.11E-3 to 6.52E-3, about a factor 2x.

Based on this, applying an elastomer coating to the deforming areas of a resonating structure could help reduce the resonant amplitude by a factor 2x at least. To push this possibility further, more testing should be done to determine the best elastomer type and thickness to increase damping as needed.

The same sample tested at 77K also showed a significant damping ratio increases on the first attempt, but this quickly degraded over the next attempts until the damping ratio became equal to regular stainless steel damping ratio at 77K. This is suspected to be because of elastomer layer delamination which was visible on the sample surface after cryogenic testing.

### 3.3 Suggestions for future setup improvement

As previously discussed in section 2, steps were taken to minimize the effect or air viscous damping and clamp frictional damping on the results, but some influence is unavoidable.

Ref. [8] describe a similar setup used in air and under vacuum and reports no significant difference between air and vacuum testing.

To try to estimate the effect of air on the results reported, the lowest structural damping coefficient measured with this setup was 9E-4. So, it can be concluded that the effect of air viscous damping and frictional damping combined, is less than 9E-4 in our setup configuration (probably significantly less- given that the material structural damping is also contributing).

To fully separate structural damping from viscous air drag damping effect, for example for the accurate design of accelerators supporting structures in vacuum (magnet cryostat supports -see Ref. [6] or SRF cavities supports) it would be best to design a testing setup under vacuum.

An example of simple and effective setup is described in Ref. [8] where the sample is enclosed in a vacuum test chamber submerged in cryogen. The sample excitation can be done through an electromagnet. Using thin blades samples instead of a rod will help simplify and improve the fixed-end clamping – where any possible friction with the sample is to be avoided – and avoid exciting resonance mode in directions other than the direction of interest.

### Conclusion

A relatively simple test setup was used to estimate structural damping values for different materials. The results obtained match relatively well with litterature data and can be used to approximate the resonnance amplitude of vibrating structures.

### References

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# Annex 1 - Evolution of the broadband amplification factor with damping ratio

As pointed out in the report, the vibration amplification at resonance:

$$\frac{|B(\omega)|}{|A|} \propto \frac{1}{\xi}$$

However, to get the overall impact of the damping ratio on amplification integrated over a range of frequency we can look at the integrated amplification factor from a unit broadband excitation. For example, looking at a unit excitation (|A|=1) from 1 Hz to 100 Hz of a system with a resonance at 50 Hz and varying the damping ratio ( $\xi=[0.001;\ 0.005;\ 0.01]$ :

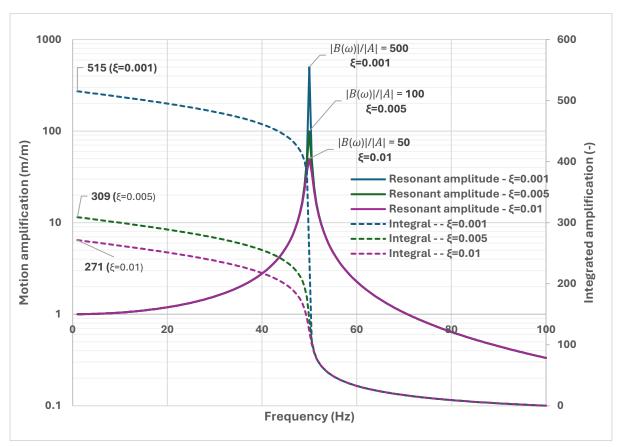


Figure 4 Amplification factor and integrated amplification from 1 Hz to 100 Hz of a unit base excitation

 $Figure\ 4\ shows\ that\ the\ integral\ of\ amplified\ motion\ from\ a\ unit\ broadband\ excitation\ is\ not\ linear\ and\ tapers\ off\ with\ increasing\ damping\ ratio.$ 

Figure 5 shows the plot of 
$$\int_{1\ Hz}^{100\ Hz} rac{|B(\omega)|}{|A|}\,df$$
 against  $\xi$  :

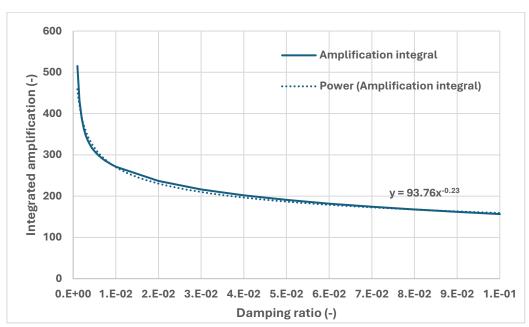


Figure 5 Integrated amplification from 1 Hz to 100 Hz against damping ratio

In this range, the interpolation points at a dependence in:

$$\int_{1\,Hz}^{100\,Hz}\!\frac{|B(\omega)|}{|A|}df \propto \frac{1}{\xi^{0.23}} \; (approximately)$$

For example, in this case, multiplying  $\xi$  by a factor 10x decreases the integrated displacement amplification only by a factor x0.59 (41% reduction in integrated displacement).

Indeed, it is obvious when looking at figure 4, that the damping ratio only really affects a very narrow band of frequencies around the resonance frequency, while the amplification at frequencies even slightly away from the exact resonance remain mostly unaffected.

Note: the limit on significance of this simplified analysis is that ground motion is usually not a consistent broadband excitation – instead ground motion amplitude decreases sharply with increasing frequencies.

# Annex 2 - Material certificates for samples used

### Ti-6Al-4V sample



Perryman company 625 Technology Dr., Coal Center, PA 15423 phone: 724-746-9390 fax: 724-746-9392

Perryman Company 213 Vandale Drive

# CERTIFICATE OF TEST

For TITANIUM INGOT

WORK ORDER 055633 HEAT NUMBER PV10144

					QUA	NTITY	PCS				
					12460	LBS	11				
N	FOR	M: Ingot,	Grade: Ti	6AI/4V, Siz	e: 30						
	BILITY										
0								_			
				Chem	nistry (W	T%) Top					
N	C	Fe	Al	v	Si	Cu	Pd	Y	H	В	Rt
.010	.023	. 22	6.08	3.97	.019	<.004	<.005	<.005	.0006	<.003	<.005
				Chemis	try (WT	%) Botto	m				
N	С	Fe	Al	v	Si	Cu	Pd	Y	H	В	R
.011	.018	.16	6.19	3.81	.012	<.004	<.005	<.005	.0005	<.003	<.00
					Comme	nts					
Proce	288:										
AR											
֡֡֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜֜	ON CAPA 6  N.010  N.011	N C .010 .023  N C .011 .018	ON CAPABILITY  6  N C Fe .010 .023 .22  N C Fe .011 .018 .16	N C Fe Al .010 .023 .22 6.08  N C Fe Al .011 .018 .16 6.19	ON CAPABILITY 6  Chem N C Fe Al V .010 .023 .22 6.08 3.97  Chemis N C Fe Al V .011 .018 .16 6.19 3.81	N FORM: Ingot, Grade: Ti6Al/4V, Size: 30  ON CAPABILITY 6  Chemistry (W  N C Fe Al V Si  .010 .023 .22 6.08 3.97 .019  Chemistry (WT  N C Fe Al V Si .011 .018 .16 6.19 3.81 .012  Comme	ON CAPABILITY 6  Chemistry (WT%) Top  N C Fe Al V Si Cu .010 .023 .22 6.08 3.97 .019 <.004  Chemistry (WT%) Botton  N C Fe Al V Si Cu .011 .018 .16 6.19 3.81 .012 <.004  Comments  Process:	N   FORM: Ingot, Grade: Ti6Al/4V, Size: 30	N FORM: Ingot, Grade: Ti6Al/4V, Size: 30  ON CAPABILITY 6  Chemistry (WT%) Top  N C Fe Al V Si Cu Pd Y .010 .023 .22 6.08 3.97 .019 <.004 <.005 <.005  Chemistry (WT%) Bottom  N C Fe Al V Si Cu Pd Y .011 .018 .16 6.19 3.81 .012 <.004 <.005 <.005  Comments	N FORM: Ingot, Grade: Ti6AI/4V, Size: 30  ON CAPABILITY 6  Chemistry (WT%) Top  N C Fe Al V Si Cu Pd Y H .010 .023 .22 6.08 3.97 .019 <.004 <.005 <.005 .0006  Chemistry (WT%) Bottom  N C Fe Al V Si Cu Pd Y H .011 .018 .16 6.19 3.81 .012 <.004 <.005 <.005 .0005  Comments	N FORM: Ingot, Grade: Ti6Al/4V, Size: 30  ON CAPABILITY 6  Chemistry (WT%) Top  N C Fe Al V Si Cu Pd Y H B  .010 .023 .22 6.08 3.97 .019 <.004 <.005 <.005 .0006 <.003  Chemistry (WT%) Bottom  N C Fe Al V Si Cu Pd Y H B  .011 .018 .16 6.19 3.81 .012 <.004 <.005 <.005 .0005 <.003

Calculated Beta Transus T/B C: 990C/998C

Residual Elements, each < .05% Residual Elements, total < .20%

Balance Titanium.

Melted in the United States of America.

The material is free of radioactive and mercury contamination.

Oxygen/Nitrogen tested in accordance with ASTM E1409.

Carbon tested in accordance with ASTM E1941.

Hydrogen tested in accordance with ASTM E1447.

B tested in accordance with ASTM E2994.

Metallics Tested By XRF in accordance with ASTM E539.

Elements not required by specification are included in Others(each) and Others(Total)

ACCREDITED

Materials Testing Laboratory

This material conforms to the chemistry and quality system requirements of the specifications and/or Purchase Order referenced. The results relate only to the lot tested and the results are contained in the records of Perryman Company.

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Michael DeAngelis Senior Laboratory Technician

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#### ABNAHMEPRUEFZEUGNIS

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REFER TO PAGE 1	W41346	SEE PG 1	SEE PG 1

HEAT NUMBER / SCHMELZE-NR. / N° DE COULEE :

592746

PRODUCT DESCRIPTION: GALL-TOUGH PLUS STAINLESS/UNS S21800 AMMEALED GROUND

ASME-SA276 UNS S21800 2019 EDITION
ASTM-A479/A479M-21 S21800 (11/01/21)
ASTM A580/A580M-18 "CHEMISTRY ONLY"
ASTM A276/A276M-17 UNS S20162/S21800
AMS 2750 REV E (07/01/12)
GE S-SPEC-35 (S-400) (02/28/21)
AMS 5848 REV E (07/ /18)
ASTM F899-20 UNS S21800
GE S-SPEC-1 (S-1000) (04/01/21)
ASTM-A314-19 UNS S21800

STM-A314-19 UNS S21800

10001517 GALL-TO BAR REV 0 (03/16/21)

SIZE 0.250000 IN.( 6.35 MM) RD

HEAT CHEMISTRY (WT%):

(TEST METHOD IS SHOWN IN PARENTHESIS)

C (COM) MN(XRF) SI(XRF) S (COM) CR(XRF) P (XRF) 0.09 7.46 17.64

NI(XRF) MO(XRF) CU(XRF) CO(XRF) N (FUS) 0.56 8.35 0.13 0.06 0.16

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GEAG SUPPLIER CODE - 21100/CARPENTER, T9406/ATHENS, 51904/LATROBE INTERGRANULAR CORROSION TESTED TO ASTM-A262, PRACTICE E - ACCEPTABLE. \*\*\*AS RECEIVED\*\*\* (DUAL) ANNEALED MICROSTRUCTURE FREE FROM CONTINUOUS GRAIN BOUNDARY CARBIDE

PRECIPITATION (NETWORK).

PRECIPITATION (NETWORK).
COUNTRY OF ORIGIN: USA
MATERIAL SUPPLIED CONFORMS TO THE TECHNICAL REQUIREMENTS OF THIS
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592746

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FURNACES WERE CALIBRATED IN ACCORDANCE WITH THE LATEST REVISION OF AMS 2750 THIS HEAT MELTED BY THE ELECTRIC ARC/AOD PROCESSES

MILL HEAT TREATMENT:

MILL HEAT TREATMENT:

TYPE SOLUTION ANNEAL
TEMP 1951F ( 1066C)
TIME (BATCH FURNACE) .50 HOURS
QUENCH WATER

HARDNESS AS SHIPPED, HEW

214

(MIDRADIUS)

THE INDENTATION MEASURING DEVICE WAS A TYPE A.

(T)RANSVERSE (L)ONGITUDINAL YIELD STRENGTH, (0.20 %) KSI(MPA) TENSILE STRENGTH, KSI(MPA) ELONGATION IN 1.05", % REDUCTION OF AREA, % 72.0( 496) 119.0( 820) 74.0 72.0

\*-->TESTED AT READING

GRAIN SIZE PER ASTM E112: 5 (KALLINGS NO.2)
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Certificate No: 3434 / 2022 Rev. 1

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USA - 02864 Cumberland, RI

USA

Material:

Dilaton36

Mat.-No.:

1.3912

UNS K 93603

Form of delivery:

Dimension:

dia. 6,350 mm dia. 0.250"

Condition:

Specification:

Order-No.: 206504 / 25.07.2022

80162691 / 1

ASTM F 1684-06 (Reapproved

2016)

Confirmation-No.:

UNS K 93603

SAE AMS-I-23011C-17 CI.

7,(chem.only)

cold drawn, annealed, in cut length

Net Weight: 503,50 kg 1110,03 lbs

No. of pieces:

20 bundles

Heat-No.: 52024

## MEASURED VALUES

Melting Composition (mass-%):

**AI** <0,01

0,01

Cr 0.04

**Mg** <0,002

Mn 0.30

36.1

< 0.003

0,004

0,10

<0,005

Tensile strength [PSI] Coefficient of thermal expansion

30 - 150 °C [10exp.-6 / K]

67900

1,8

Cert Rev. 1 by A.Goersch dt. 05/04/2023 (original cert dt. 10/27/2022):

"UNS number corrected"

The material supplied is free from contamination of mercury and has not been weld repaired.

The material was manufactured without the use of ozone-depleting chemicals.

The reported results represent the actual attributes of the material furnished and indicate full compliance with all applicable specification and contract requirements.

The recording of false, fictitious or fraudulent statements or entries on this document may be punished as a felony under Federal Statutes.

DFAR 252-225-7009 Compliant Country of Origin: Germany

# **Steel C52100**



Banner Service Corp www.bargrind.com 494 E Lies Road Carol Stream, IL 60188

# **Certificate of Compliance**

Customer Number	C000539	Customer Name	Castle Metals IL	
Customer PO	637232	Customer Address	3800 Enterprise Drive	
Customer Part Number	2500/.2450" E52100		Janesville, WI 53546	
Producing Plant	Carol Stream	Ship Date	06/09/2023	
Item Number	C52100 R 0.2500	Production Order Nu	mber	
Item Description	CN52R.016- 1/4" RD E52100 .2500/.2450" CF X 144" ASTM.A295	Sales order		SO-C0072625
		Heat Number		418531
		Quantity in lb		820.00
		Quantity in Pounds		820.00

### **Heat Attributes**

Heat Attribute Description	Result		Heat Attribute Description		Result	
Physical		- 11	Chemical			
OD for Round, Thickness for Flat	0.2500	- 11	Chromium		1.4200	
Grade	C52100	- 11	Copper		0.0900	
Length	144.0000	- 11	Manganese		0.2900	
Shape	R		Molybdenum		0.0500	
Chemical		- 11	Phosphorous		0.0040	
Tin	0.0100		Sulfur		0.0050	
Titanium	0.0010		Silicon		0.2500	
Hydrogen	0.0001		Calcium		0.0002	
Antimony	0.0010		Mechanical			
Aluminum	0.0380	- 11	Cast Method		INGOT	
Nickel	0.0800		Rolling/Processing Source 1		FN STEEL- NETHERLANDS	
Oxygen	0.0003		Rolling/Processing Source 2		SULLIVAN STEEL- US	
Lead	0.0010		Annealed		Υ	
Arsenic	0.0100	- 11	Grain Size		8.0000	
Aluminum	0.0380	- 11	Melt Source/Location		OVAKO-SWEDEN	
Carbon	0.9900		UltimateTensile(KSI)		90.0680	

# MERCURY FREE

Unless noted, this material meets RoHS compliance, REACH compliance, and Conflict Minerals Restrictions.

BANNER SERVICE CORP CERTIFIES THAT THE ABOVE INFORMATION IS TAKEN FROM A CHEMICAL AND/OR MECHANICAL TEST REPORT FURNISHED TO US BY OUR SUPPLIER AND THAT SUCH TEST REPORT IS ON FILE IN OUR OFFICE.

David Hallin, Quality Manager Printed Date 06/09/2023

### Aluminum 6061-T6





### **CERTIFIED TEST REPORT**

http://Online.KaiserAluminum.com

CUSTOMER PO NUMBER:	TOMER PO NUMBER: CUSTOMER PART NUMBER:			PRODUCT DESCRIPTI	PRODUCT DESCRIPTION:				
660929-1 7891				Cold Finished Ro	Cold Finished Round				
KAISER ORDER NUMBER:	ISER ORDER NUMBER: LINE ITEM: SHIP		SHIP DATE:	ALLOY:	TEMPER				
741256	1		02/22/2021	6061	т6				
WEIGHT SHIPPED:	QUANTITY:		B/L NUMBER:	DIAM/DAF/THKNS:	WIDTH:	LENGTH:			
1085 lbs. 1576 pcs.			45813	0.250 in.	0.000 in.	144.0 in.			
SHIPTO: A M Castle & Co 3800 Enterprise Drive Janesville, WI 53546 USA				1420 Kensington Suite 220	SOLD TO: A M Castle & Co 1420 Kensington Road				

### Test Code: 1000 AC

### **Actual Physical Properties**

# Chemical Composition, WT. % (Aluminum Remainder)

	Si	Fe	Cu	Mn	Mg	Cr	Zn	Ti	OTHERS EACH	OTHERS TOTAL	Lot
Actual									2.25	0.15	10079543
Max Min	0.40		0.40			0.35	0.25	0.15	0.05	0.15	

### Specifications

ISO 9001, ASTMB211-19, AMS4117L, AMS-QQ-A-225/8A, A96061\_01\_R20 w exceptions

### CERTIFICATION

Kaiser Aluminum Fabricated Products, LLC ("Kaiser") hereby certifies that the metal shipped under this order has been inspected and tested and found in conformance with the applicable specifications forming a part of the description set forth in Kaiser's sales acknowledgement form. Any warranty is limited to that shown on Kaiser's general terms and conditions of sale. Test reports are on file, subject to examination.

John Rennekamp, Quality Manager

Page 1 of 2



Plant Serial: 282834 Kaiser Order Number: 741256

Line Item: 1

# Carbon fiber sample

Length 12"

Materials Standard Modulus Carbon

OD 0.25

Outside Dimension 0.25"

Shape Round Rod

Weight 0.246 lb

# **Technical Data**

# **Minimum Properties:**

Tensile Strength: 250 ksi | 1.72 GPa
Tensile Modulus: 20 msi | 138 GPa
Ultimate Shear Strength: 6.0 ksi | 41.3 Mpa
Ultimate Tensile Strain: 1.5%
Flexural Strength: 265 ksi | 1.83GPa
Flexural Modulus: 19 msi | 131 GPa
Coefficient of Thermal Expansion: -0.1 ppm/°F | -0.2ppm/ °C
Density: 0.054lbs/in³ | 1.5g/cm³

Fiber Volume: 62%
Glass Transition Temperature: 100°C
Matrix Material: Bisphenol Epoxy Vinyl Ester

# **Tolerances**

Diameter Tolerance: +/- 0.007"

# Ultem 2300 sample



Customer	CURBELL PLASTICS INC		
Customer Location	ROCHESTER, NY 14624	Ensinger Order#	929638
Customer Order#	4500565058	Order Line #	5
Certificate	AZE933	Issued	03/19/2025

Ensinger, Inc. certif	les the material shipped is the following		1	
Trade Name	TECAPEI™GF30	Size / Shape	0.25"	ROD
Color	NATURAL	Nominal Length	120.0000"	
Material	GF POLYETHERIMIDE (PEI)	Nominal Width	N/A	
Production Lot(s)	424127	Quantity	10.0000	FT
Resin Lot(s)	1002402748	Country of Origin	USA	
Base Resin	Ultem™ 2300 Series	Manufacture Date	May 2022	

Ensinger, Inc. certifies typical production lots of the above identified material have been found to comply with the following Specifications and/or Compliances

Base Resin Spec ASTM D5205-24 PEI0110G30A96299 159 MPa 224 MPa 208C

Shape Spec

ASTM D7293-19 S-PEI0213

Resin Agency Compliances Ref. UL File #E121562 for Resin & Flammability

FDA 21 (

Prop 65

FDA 21 CFR 177.1595

WARNINGI This product contains Prop 65 listed substances, Contact compliance@ensingerusa.com for a complete declaration,

Customer Compliances

Comments

Every batch of material is not tested. If additional testing is required, it must be requested at time of order.

Reported resin lot is representative of the raw material used to manufacture this product.

Ranald Woolley

Ronald Woolley Quality Assurance Production Lot(s) Resin Lot(s)



